

ES100 March 29, 1999 E. F. Thacher

## Topics

- Reducing the airfoil lift data (CLFIT.M)
- Making decisions, part I
- Ground vehicle in steady motion
- Some vehicle geometry
- Lift and drag
- Vehicle wind tunnel testing
- Rolling resistance
- Making decisions, part II
- Coast-down tests; reducing data (COAST.M)

### Making Decisions I

#### Relational operators

Find where the angle of attack is not greater than 21 degrees. Then fit a straight line to the data.

z = ( angle <= 21 )

 Array z has 1's where true, zero's where false

#### CLFIT.M I



### CLFIT.M II

%	Lc	bad	and	parse	tł	ıe	data	fi	Le		
10	load a:\ld1.txt										
V	=	ld1	L(:,4	4);	%	ve	ector	of	air	sp	eeds
					%	( m /	/s)				
L	=	1d1	L(:,2	2);	%	Ve	ector	of	lif	ts	(N)
D	=	1d1	L(:,:	3);	%	Ve	ector	of	drag	gs	(N)
angle = ld1(:,1);% v						Ve	ector	of	ang	les	of
					%	at	tack	: <b>(</b> N	)		
%	Aj	lrfc	oil d	limens	ior	1S					
C	=	0.1	L0152	2;	%	m					
b	=	0.2	254;		%	m					
S	=	b*c	2;		%	p	lanfc	orm a	area	<b>(</b> m	L^2)

#### CLFIT.M III

% Calculate air density
rho = airden ( p, T );

#### CLFIT.M IV

- CL = L./(S\*q);
- % Locate angles <= 21 degrees</pre>

#### z = ( angle <= 21 );</pre>

fprintf( '\nThe array z:\n %g\n', z)
limit\_index = sum(z); %count the 1's

#### CLFIT.M V

- % Fit a straight line to the data subset
- L = polyfit( angle(1:limit\_index),...

cL(1:limit\_index), 1 );

fprintf( '\nThe slope is %g ;

- the intercept is  $g \ \ L(1), L(2)$
- % Predict cL at and + angles
- x = [-angle(limit\_index)-6:1:... angle(limit\_index)+6];
- y = polyval( L, x );

#### CLFIT.M VI

% Plot cL and cL fit vs. angle of attack
plot ( angle, cL, 'square', x, y, '-' );
legend ('lift coefficient',...
 '10-point linear fit', 0 );
xlabel ('angle of attack (degrees)');
ylabel ('lift coefficient');
title ('Lift Coefficient and Linear Fit vs.
 Angle of Attack')

grid

Demonstrate CLFIT.M

• Note: the prediction

$$c_L(-\alpha) = -c_L(\alpha)$$

is true.

Neglect transverse forces and moments
 x
 x
 x

Neglect transverse forces and moments
 x
 x
 x

Neglect transverse forces and moments
 x , f , drag
 λ , drag

















Summing Forces in x- and z-directions:

## $\sum F_x = T - R - D - W \sin \lambda = 0$ $\sum F_z = L + N - W \cos \lambda = 0$

$$R = R_1 + R_2$$
$$N = N_1 + N_2$$

#### Accelerating in X-Direction

$$T - R - D - W\sin\lambda = M_e \frac{dV}{dt}$$
$$L + N - W\cos\lambda = 0$$

The effective mass,  $M_e$ , accounts for the rotational inertia of the wheels. It is about 1.03 M, typically.

#### Some Vehicle Geometry



"drag area" is c<sub>D</sub>A<sub>D</sub>, [m<sup>2</sup>]

$$c_L = \frac{L}{A_D \frac{1}{2}\rho V^2}$$

$$\frac{N}{(m^{2})\left(\frac{N}{m^{2}}\right)} \Rightarrow c_{L} \text{ is dimensionless}$$



$$\frac{N}{(m^{2})\left(\frac{N}{m^{2}}\right)} \Rightarrow c_{L} \text{ is dimensionless}$$









#### Drag Coefficient of Car



### Lift, Drag in Straight Ahead Flow

#### Lift and drag depend on

- The relative airspeed
- The viscosity and density of the air

#### The shape and smoothness of the vehicle

- Including wheels and wheel wells, and the

- Ventilation system

- The pitch angle of the vehicle
- The proximity of the ground

## Measuring $c_L$ and $c_D$ in Tunnel



### Measuring $c_L$ and $c_D$ in Tunnel



#### Utility of Model Tests

Use c<sub>L</sub> and c<sub>D</sub> for full scale car
 Geometrically similar car
 Same Reynolds number

However, drag of actual car usually greater than model

- Model usually simplified
- Some full-scale test facilities exist

**Reynolds Number** 



### **Rolling Resistance**

Contact patch force
Proportional to N
Bearing friction (small)
Proportional to V and W
Wheel rotational drag (smallest)
Proportional to V

$$R = \mu W = (\mu_1 \cos \lambda + \mu_2 V)W$$

Coast-Down Test to Measure  $c_D A_D, \mu_1, \mu_2$ 

- Measure W, p (pressure), and T (temp. Celcius)
- No wind, horizontal road (ideally)
- Initial speed greater than about 40 mph
- Coast down in straight line and record V at intervals
- [Note: μ means "rolling resistance coefficient," in this case -- not viscosity]

#### Reducing Coast Down Data

• Fit V(t), find V(t), fit a quadratic in V to it:

$$\dot{V}(t) = a_2 V^2 + a_1 V + a_0$$

• Where

$$a_{2} = \frac{c_{D}A_{D}\rho}{2M_{e}}, \quad a_{1} = \frac{\mu_{2}W}{M_{e}}, \quad a_{0} = \frac{\mu_{1}W}{M_{e}}$$

#### Making Decisions II

If the user selects numeric derivative
...then do that method and go on
Else she wants to smooth the data first
...then do that and go on
End of decision

#### First Derivative Approximations

#### **Forward difference**

$$\frac{dV}{dt} \approx \frac{V_{t+\Delta t} - V_t}{\Delta t} = \frac{V_{i+1} - V_i}{\Delta t}$$

Central difference

$$\frac{dV}{dt} \approx \frac{V_{i+1} - V_{i-1}}{2\Delta t}$$

**Backward difference** 

$$\frac{dV}{dt} \approx \frac{V_i - V_{i-1}}{\Delta t}$$

#### Numerical Derivatives II



#### COASTD.M Flow Chart I



#### Numerical dV/dt



### COASTD.M I

%	LC	bad	and	pars	se the	data	a fi	le				
lc	ac	l a:	\cdd	lata	-ascii	.;						
t	=	1d1	.(:,1	L);	ନ୍	ve	ctor	of	time	25	(sec)	)
V	=	1d1	.(:,2	2);	ନ୍	s ve	ctor	of	car	sp	eeds	at
					9	ti.	nes	(m/s	<b>z</b> )			

```
% Car mass and profile area
load a:\cdtest -ascii;
Ap = cdtest(2); % m^2
M = cdtest(5); % kg
p = cdtest(6);
Tc = cdtest(6);
T = tk (Tc);
rho = airden (p*1000, T);
```

#### COASTD.M II

#### The "choice" if-block

If choice == 1

## [un-smoothed data, numeric derivative statements]

else

## [cubic-polynomial fitted to data, derivative is derivative of polynomial]



### COASTD.M III

- % Start of "choice" if-block
  if choice == 1
  - % Un-smoothed data, numeric derivative n = length(V); clear dVdt dt = t(2)-t(1);
  - % Loop through all data points

#### [next slide]

#### [if choice == 1, continued]

for i = 1:n**if** i == 1 dVdt(i) = (V(i+1)-V(i))/dt;elseif i == n dVdt(i) = (V(i)-V(i-1))/dt;else dVdt(i) = (V(i+1)-V(i-1))/(2\*dt);end end % End of loop through data % Reduce the results a = polyfit (V, dVdt', 2);[cDm mulm mu2m] = reduce (a, cdtest);

#### COASTD.M IV

#### else

% Try cubic smoothing b3 = polyfit (t, V, 3); $Vf3 = b3(1)*t.^3 + b3(2)*t.^2 + ...$ b3(3)\*t + b3(4);dVdt3 = 3\*b3(1)\*t.^2 + 2\*b3(2)\*t +... b3(3); % Reduce the results a3 = polyfit (Vf3, dVdt3, 2);[cDm mulm mu2m] = reduce (a3, cdtest); end % End of "choice" if-block

#### COASTD.M V

- % Print results
  fprintf ('\nResults:\n\tcD = %g\n
   \tmu1 = %g\n\tmu2 = %g (s/m)\n\n',...
   cDm, mu1m, mu2m);
- % Calculate errors
  errors (cDm, mu1m, mu2m, cdtest);

#### New Functions in COASTD.M I

function errors ( cD, mu1, mu2, cdtest )

ecD = abs( (cD/cdtest(1)) - 1 )\*100; emu1 = abs( (mu1/cdtest(3)) - 1 )\*100; emu2 = abs( (mu2/cdtest(4)) - 1 )\*100; fprintf ('\n\tError in cD = %g %%\n \tError in mu1 = %g %%\n \tError in mu2 = %g %%\n\n',... ecD, emu1, emu2)

#### New Functions in COASTD.M II

function [cD, mu1, mu2] = reduce(a, cdtest); % Assign convenient names AD = cdtest(2);M = cdtest(5);p = cdtest(6);Tc = cdtest(7);% Calculate T, rho, Me, and W T = tk (Tc);rho = airden (p\*1000, T);Me = 1.03\*M;W = M\*9.807;

#### Function REDUCE.M, cont.

%	Find	cD,	mul,	mu2.	Recall	that:
%	a(	(1) =	<b></b> 5 <sup>3</sup>	*cD*AI	)*rho/Me	;
%	a(	(2) =	-mu2	2*W/Me	2;	
%	a(	(3) =	-mul	1*W/Me	2;	
cI	) = -a	a(1)*	2*Me,	/(AD*r	cho);	
mu	ı1 = -	-a(3)	*Me/V	N;		
mu	12 = -	a(2)	*Me/V	N;		

#### Results from COASTD.M

# Effect of number of measurements Effect of standard deviation